

Sliding Mode Control for High-Precision Motion of a Piezostage

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Abstract—In this paper, control of piezostage using sliding mode control (SMC) method is presented. Due to the fast dynamics of the piezostage and since high accuracy is required the special attention is paid to avoid chattering. The presence of hysteresis characteristics represents main nonlinearity in the system. Structure of proposed SMC controller is proven to offer chattering-free motion and rejection of the disturbances represented by hysteresis and the time variation of the piezostack parameters. In order to enhance the accuracy of the closed loop system, a combination of disturbance rejection method and the SMC controller is explored and its effectiveness is experimentally demonstrated. The disturbance observer is constructed using a second-order lumped parameter model of the piezostage and is based on SMC framework. Closed-loop experiments are presented using proportional-integral-derivative controller and sliding mode controller with disturbance compensation for the purpose of comparison.

Index Terms—Discrete-time control, high-precision motion, piezostage, sliding mode control (SMC).

I. INTRODUCTION

PIEZOELECTRIC actuators have shown a great potential in applications that require submicrometer down to nanometer motion. The advantages that piezoelectric actuators offer are the absence of friction and stiction characteristics that exist in other actuators. Thus, piezoelectric actuators are ideal for very high-precision-motion applications. The main characteristics of piezoelectric actuators are: extremely high resolution in the nanometer range, high bandwidth up to several kilohertz range, a large force up to few tons, and very short travel in the submillimeter range (see [1]). Application areas of piezoelectric actuators include: micromanipulation, microassembly, add-ons for high-precision cutting machinery, and as secondary actuators in macro/micromotion systems such as dual-stage hard-disk drives. In all of these applications, the accuracy of positioning is very important and in many cases the closed loop control is the only answer. Despite this, there are many attempts (see [2] and [3]) to drive piezoelectric actuators as an open loop system with fine compensation of the hysteresis nonlinearity in one or another way. With development of accurate position transducers, the possibility to use robust feedback-based non-

linear control methods is becoming an attractive alternative to the model-based compensation.

Despite the fact that a piezoelectric actuator is a distributed parameters system, modeling for control purposes is based on a lumped parameters system. It is possible to drive piezoelectric actuators with either voltage or charge as input. The former is easier to implement in hardware and is the most common mode of controlling these actuators. However, a piezoelectric actuator driven by voltage as input will exhibit nonlinearity between the input (voltage) and output (position). This nonlinearity is mainly due to the parasitic hysteresis characteristics of piezoelectric crystals. It has been shown in many other works (see [2]) that hysteresis behavior does not exist in the case of a piezoelectric actuator driven by charge and that the actuator exhibits almost linear behavior between charge and position. However, as mentioned before, hardware realization of charge controllers is very difficult and voltage supply-based control is mostly preferred.

A major difficulty in using piezoelectric actuators is the hysteresis effect, which causes large positioning errors. There are many techniques used in order to handle the nonlinearities brought by this effect such as feedback and model-based feedforward control. Also, in [4], iterative method is used in order to find the hysteresis that compensates feedforward input for high-precision positioning. In addition to the hysteresis characteristics, piezoelectric actuators also have dynamic creep effect that has to be taken into account. In [5], both the hysteresis and dynamic creep effects are given importance and operator-based inverse feedforward controller is applied. It has been shown that this controller works well for highly dynamic operation and that it is simple and inexpensive for mechatronic devices with hysteresis characteristics. There has been also research on the mathematical modeling of hysteresis, such as in [2], [3], [6]–[8] where new results for the modeling of physical hysteresis and its applications in dynamic research are shown. Complicated models of the hysteresis allow for accurate control of these actuators but are limited due to presence of other internal disturbances such as creep. In [2], complex and accurate model of hysteresis is presented, but is hard to implement and too complex for control applications. In [3], [6], and [7], simpler models of hysteresis are proposed, however, those models fail to precisely represent hysteresis behavior throughout the whole range of input voltage of the piezoelectric actuator. The problem of hysteresis was also approached by using neural-network (NN) technology. In [9], they trained a recurrent NN to mimic the behavior of inverse characteristic of the piezocrystal and they used this trained network in series with the piezoactuator.

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90 Use of a hysteresis model provides some advantages; it does
 91 not need the measurement of the mechanical coordinates and
 92 is helpful in applications where the use of sensors for position
 93 measurement is impractical.

94 In [7], H_∞ -based closed-loop control is presented with
 95 model-based hysteresis compensation. While the method pro-
 96 duces good results, it can be made simpler if the hysteresis
 97 model-based compensation is replaced with a simpler method-
 98 ology. In [10], a NN-based feed-forward assisted proportional-
 99 integral-derivative (PID) controller was proposed. A hybrid
 100 control strategy using a variable structure control is suggested
 101 for submicrometer positioning control [9], [11]. These methods
 102 need an explicit system model for the control design, and
 103 the performance achievable depends on the accuracy of the
 104 model. In [14], a sliding-mode approach for linear discrete-time
 105 systems is proposed. Based on the proposed method in [14] and
 106 [17], $O(T_s^2)$ bound of the sliding surface is achieved. In this
 107 paper, we claim the same accuracy, but, with partial knowledge
 108 of system dynamics.

109 In this paper, the aim is to design a motion controller for
 110 piezostage having position sensor based on the assumption that
 111 the piezostage can be modeled as a linear lumped parameters
 112 ($T, m_{\text{eff}}, c_{\text{eff}}, k_{\text{eff}}$) second-order electromechanical system with
 113 voltage as the input and position as the output coordinate and
 114 hysteresis nonlinearity being the major disturbance effecting
 115 the system. Furthermore, it is assumed that the parameters of
 116 the model are bounded and have some so-called nominal values
 117 (T_N, m_N, c_N, k_N).

118 In this paper, the sliding mode methods are applied in the
 119 design of a high-accuracy piezoactuator position. The solution
 120 proposed here combines the sliding mode controller and the
 121 disturbance rejection method in order to achieve high accuracy
 122 in the actuator trajectory tracking. For the disturbance estima-
 123 tion, a sliding mode observer-based disturbance compensation
 124 method is used here. By manipulating model of a piezoactuator
 125 in a form where nonlinearities due to hysteresis are presented
 126 as an additive disturbance acting together with external force
 127 to the mechanical system a simple second-order observer is
 128 designed to estimate lumped disturbance.

129 This paper is organized as follows. In Section II, a suit-
 130 able model of a piezoactuator, based on already known re-
 131 sults, is presented. In Section III, the sliding mode-based con-
 132 troller and in Section IV the observer design is presented. In
 133 Section V, experimental results verifying theoretical works
 134 are presented.

135 II. MODEL OF THE PIEZOSTAGE

136 In this paper, a piezostage that consists of a piezodrive
 137 integrated with a sophisticated flexure structure for motion
 138 amplification is used. The flexure structure is wire-EDM-cut
 139 and is designed to have zero stiction and friction. Fig. 1 shows
 140 the piezodrive integrated flexure structure.

141 In addition to the absence of internal friction, flexure stages
 142 exhibit high stiffness and high load capacity. Flexure stages
 143 are also insensitive to shock and vibration. However, since the
 144 piezodrive exhibits nonlinear hysteresis behavior, the overall
 145 system will also exhibit the same behavior.

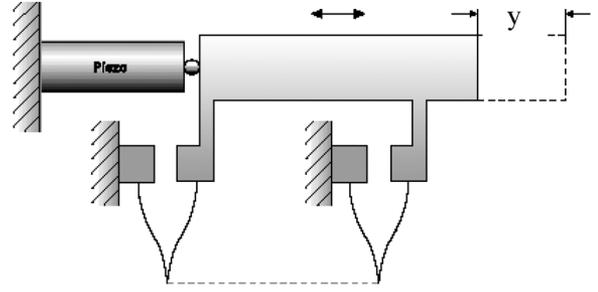


Fig. 1. Structure of a flexure piezostage.

The dynamics of the piezostage can be represented by the following second-order differential equation coupled with hysteresis in the presence of external forces

$$m_{\text{eff}}\ddot{y} + c_{\text{eff}}\dot{y} + k_{\text{eff}}y = T(u(t) - h(y, u)) - F_{\text{ext}} \quad (1)$$

where m_{eff} denotes the effective mass of the stage, y denotes the displacement of the stage, c_{eff} denotes the effective damping of the stage, k_{eff} denotes the effective stiffness of the stage, T denotes the electromechanical transformation ratio, u denotes the input voltage and $h(y, u)$ denotes the nonlinear hysteresis that has been found to be a function of y and u , [2], and F_{ext} is the external force acting on the stage.

The model represented by (1) is found from the work of [2] and it shows that from the mechanical motion the hysteresis may be perceived as a disturbance force that satisfies matching conditions. This means that the sliding mode-based control should be able to reject the influence of the hysteresis nonlinearity on the mechanical motion. At the same time, it is obvious that the lumped disturbance consisting of the external force acting on the system and the hysteresis can be estimated, thus allowing the application of the disturbance rejection method in the overall system design.

166 III. SLIDING-MODE-CONTROLLER DESIGN

167 A. Controller Design

To facilitate the derivation of the control law, (1) is written into the state-space form

$$\begin{aligned} \dot{x}_1 &= \dot{y} = x_2 \\ \dot{x}_2 &= \ddot{y} = -\frac{k_{\text{eff}}}{m_{\text{eff}}}x_1 - \frac{c_{\text{eff}}}{m_{\text{eff}}}x_2 + \frac{T}{m_{\text{eff}}}u - \frac{T}{m_{\text{eff}}}h - \frac{F_{\text{ext}}}{m_{\text{eff}}}. \end{aligned} \quad (2)$$

It is possible to write (3) in a more general form as shown below

$$\dot{x} = f(x, h, F_{\text{ext}}) + Bu. \quad (4)$$

The aim is to drive the states of the system into the set S defined by

$$S = \{x : G(x^r - x) = \sigma(x, x^r) = 0\} \quad (5)$$

where $G = [\lambda \ 1]$ with λ being a positive constant, x is the state vector $x^T = [x_1 \ x_2]$, x^r is the reference vector $(x^r)^T = [x_1^r \ x_2^r]$, and $\sigma(x, x^r)$ is the function defining sliding mode manifold.

177 The derivation of the control law starts with the selection of
178 the Lyapunov function, $V(\sigma)$, and an appropriate form of the
179 derivative of the Lyapunov function, $\dot{V}(\sigma)$.

180 For single-input–single-output systems such as (3), required
181 to have motion in manifold (5), natural selection of Lyapunov
182 function candidate seems in the form

$$V(\sigma) = \frac{\sigma^2}{2} \quad (6)$$

183 Hence, the derivative of the Lyapunov function is

$$\dot{V}(\sigma) = \sigma\dot{\sigma}. \quad (7)$$

184 In order to guarantee the asymptotic stability of the solution
185 $\sigma(x, x^r) = 0$, the derivative of the Lyapunov function may be
186 selected to be

$$\dot{V}(\sigma) = -D\sigma^2 \quad (8)$$

187 where D is a positive constant. Hence, if the control can be
188 determined from (7) and (8), the asymptotic stability of solution
189 (5) will be guaranteed since $V(\sigma) > 0$, $V(0) = 0$, and $\dot{V}(\sigma) <$
190 0 , $\dot{V}(0) = 0$. By combining (7) and (8), the following result is
191 obtained

$$\sigma(\dot{\sigma} + D\sigma) = 0. \quad (9)$$

192 A solution for (9) is as follows

$$\dot{\sigma} + D\sigma = 0. \quad (10)$$

193 The derivative of the sliding function is as follows

$$\dot{\sigma} = G(\dot{x}^r - \dot{x}) = G\dot{x}^r - G\dot{x}. \quad (11)$$

194 From (11) and using (4)

$$\dot{\sigma} = \underbrace{G\dot{x}^r - Gf}_{GBu_{\text{eq}}} - GBu(t) = GB(u_{\text{eq}} - u(t)). \quad (12)$$

195 If (12) is substituted in (10) and the result is solved for the
196 control

$$u(t) = u_{\text{eq}} + (GB)^{-1}D\sigma. \quad (13)$$

197 It can be seen from (12) that u_{eq} is difficult to calculate. Using
198 the fact that u_{eq} is a continuous function, (12) can be written in
199 discrete-time form after applying Euler's approximation

$$\frac{\sigma((k+1)T_s) - \sigma(kT_s)}{T_s} = GB(u_{\text{eq}}(kT_s) - u(kT_s)) \quad (14)$$

200 where T_s is the sampling time and $k = Z^+$. It is also necessary
201 to write (13) in discrete-time form just as it was done before

$$u(kT_s) = u_{\text{eq}}(kT_s) + (GB)^{-1}D\sigma(kT_s). \quad (15)$$

If (14) is solved for the equivalent control, the following is
202 obtained 203

$$u_{\text{eq}}(kT_s) = u(kT_s) + (GB)^{-1} \left(\frac{\sigma((k+1)T_s) - \sigma(kT_s)}{T_s} \right). \quad (16)$$

204 Since the system is causal, and it is required to avoid calculation
205 of the predicted value for σ , control cannot be dependent on a
206 future value of σ . Having equivalent control as a continuous
207 function, the current value of the equivalent control will be
208 approximated by a single time-step backward value computed
209 from (16) as follows

$$\hat{u}_{\text{eq}_k} \cong u_{\text{eq}_{k-1}} = u_{k-1} + (GB)^{-1} \left(\frac{\sigma_k - \sigma_{k-1}}{T_s} \right) \quad (17)$$

210 where \hat{u}_{eq_k} (or $\hat{u}_{\text{eq}}(kT_s)$) is the estimate of the current value of
211 the equivalent control. If (17) is substituted in (15)

$$u_k = u_{k-1} + (GBT_s)^{-1}((DT_s + 1)\sigma_k - \sigma_{k-1}). \quad (18)$$

212 Note that in certain applications where only partial state mea-
213 surements exist, observers can be used to estimate the unknown
214 states in order to compute σ_k . In this paper, the unknown state is
215 the velocity and is estimated using a discrete derivative. Hence,
216 control (18) is suitable for implementation since it requires
217 measurement of the sliding mode function and value of the
218 control applied in the preceding step. Since, the above control
219 law is derived from discrete-time approximations based on the
220 continuous-time equations. Hence, these approximations will
221 introduce errors in the control that must be analyzed carefully.

B. Closed-Loop Behavior With the Approximated Control 222

223 As a consequence of the approximations that were made in
224 the derivation of the discrete-time control law, some deviations
225 in the sliding surface from the desired sliding manifold may
226 exist. This deviation of the sliding surface from the desired
227 manifold at each sampling instant will be analyzed. Intersam-
228 pling behavior is also analyzed.

229 Considering (4), the derivative of the sliding surface is
230 given by

$$\dot{\sigma}(t) = G(\dot{x}^r - \dot{x}) = G\dot{x}^r - Gf - GBu(t). \quad (19)$$

231 The discrete-time equivalent of the sliding manifold can be
232 obtained by taking the integral on both sides of (19) from kT_s
233 to $(k+1)T_s$

$$\sigma_{k+1} - \sigma_k = \int_{kT_s}^{(k+1)T_s} (G\dot{x}^r - Gf - GBu(t)) dt. \quad (20)$$

234 Applying a sample and hold to the control input between
235 consecutive samples $u(t) = u_k$ for $kT_s \leq t < (k+1)T_s$

$$\sigma_{k+1} - \sigma_k = \int_{kT_s}^{(k+1)T_s} (G\dot{x}^r - Gf) dt - T_s GBu_k. \quad (21)$$

236 Using the assumptions that \dot{x}^r and f are smooth and bounded,
237 the integrations in (21) can be approximated by using Euler's
238 integration

$$\sigma_{k+1} = \sigma_k + T_s G(\dot{x}_k^r - f_k) - T_s G B u_k + O(T_s^2). \quad (22)$$

239 Here, $O(T_s^2)$ is the error introduced due to Euler's integration,
240 [16]. If the control defined by (18) is introduced into (22)

$$\begin{aligned} \sigma_{k+1} = & \sigma_k + T_s G(\dot{x}_k^r - f_k) - T_s G B u_{k-1} \\ & - T_s D \sigma_k - \sigma_k + \sigma_{k-1} + O(T_s^2). \end{aligned} \quad (23)$$

241 After some simplifications (23) is reduced to

$$\sigma_{k+1} = T_s G(\dot{x}_k^r - f_k) - T_s G B u_{k-1} - T_s D \sigma_k + \sigma_{k-1} + O(T_s^2). \quad (24)$$

242 If $T_s G(\dot{x}_{k-1}^r - f_{k-1})$ is added and subtracted from the r.h.s of
243 (24), the following is obtained

$$\begin{aligned} \sigma_{k+1} = & T_s G(\dot{x}_k^r - f_k) - T_s G(\dot{x}_{k-1}^r - f_{k-1}) - T_s D \sigma_k \\ & + \underbrace{T_s G(\dot{x}_{k-1}^r - f_{k-1}) - T_s G B u_{k-1}}_{\sigma_k - \sigma_{k-1} + O(T_s^2)} \\ & + \sigma_{k-1} + O(T_s^2). \end{aligned} \quad (25)$$

244 After some simplifications, (25) becomes

$$\sigma_{k+1} = \sigma_k - T_s D \sigma_k + T_s G(\Delta \dot{x}_k^r - \Delta f_k) + O(T_s^2) \quad (26)$$

245 where $\Delta \dot{x}_k^r = \dot{x}_k^r - \dot{x}_{k-1}^r$ and $\Delta f_k = f_k - f_{k-1}$. Note that if
246 $D = 1/T_s$, then the r.h.s of (26) is of order $O(T_s^2)$, keeping in
247 mind that \dot{x}^r and f are smooth and bounded. Hence

$$\sigma_{k+1} = O(T_s^2). \quad (27)$$

248 Hence, it is shown that the maximum deviation from the sliding
249 surface at each sampling instant is of order $O(T_s^2)$.

250 Next, it will be shown that the intersampling deviation of
251 the sliding surface from the desired manifold is also of order
252 $O(T_s^2)$.

253 Consider the intersampling instant of $t = kT_s + \tau$ where $0 \leq$
254 $\tau \leq T_s$. If (19) is integrated on both sides from kT_s to $kT_s + \tau$

$$\sigma(kT_s + \tau) - \sigma_k = \int_{kT_s}^{kT_s + \tau} (G\dot{x}^r - Gf - GBu(t)) dt. \quad (28)$$

255 Applying the sample and hold to the control and Euler's inte-
256 gration to the remaining integral gives

$$\sigma(kT_s + \tau) = \sigma_k + \tau G(\dot{x}_k^r - f_k) - \tau G B u_k + O(\tau^2). \quad (29)$$

257 If the control defined by (18) is introduced into (29)

$$\begin{aligned} \sigma(kT_s + \tau) = & \sigma_k + \tau G(\dot{x}_k^r - f_k) - \tau G B u_{k-1} \\ & - \tau D \sigma_k - \frac{\tau}{T_s}(\sigma_k - \sigma_{k-1}) + O(\tau^2). \end{aligned} \quad (30)$$

If $\tau G(\dot{x}_{k-1}^r - f_{k-1})$ is added and subtracted from the r.h.s of
(24) and $D = 1/T_s$, the following is obtained

$$\begin{aligned} \sigma(kT_s + \tau) = & \sigma_k + \frac{\tau}{T_s} G(T_s(\Delta \dot{x}_k^r - \Delta f_k)) - \frac{\tau}{T_s} \sigma_k - \frac{\tau}{T_s} \sigma_k \\ & + \frac{\tau}{T_s} G \underbrace{(T_s(\dot{x}_{k-1}^r - f_{k-1}) - T_s B u_{k-1})}_{\sigma_k - \sigma_{k-1} + O(T_s^2)} \\ & + \frac{\tau}{T_s} \sigma_{k-1} + O(\tau^2). \end{aligned} \quad (31)$$

Further simplifications lead to

$$\sigma(kT_s + \tau) = \sigma_k - \frac{\tau}{T_s} \sigma_k + \frac{\tau}{T_s} G(T_s(\Delta \dot{x}_k^r - \Delta f_k)) + O(\tau^2). \quad (32)$$

If \dot{x}^r and f are smooth and bounded then

$$\sigma(kT_s + \tau) = \sigma_k - \frac{\tau}{T_s} \sigma_k + O(\tau^2). \quad (33)$$

Note that if $\sigma_k = O(T_s^2)$, as was shown previously, then the
maximum intersampling value of the sliding function is $O(T_s^2)$.
Hence

$$\sigma(kT_s + \tau) = O(T_s^2). \quad (34)$$

C. Lyapunov Stability of the Closed-Loop System

In this section, it will be shown that with discrete-time
control defined by (18), it is possible to satisfy the Lyapunov
condition (10) in discrete time.

Starting with the definition of the Lyapunov function in
discrete-time, proportional to one defined by (6)

$$V_k = \sigma_k^2. \quad (35)$$

The difference of two consecutive values of the Lyapunov
function in discrete time can be given by

$$V_{k+1} - V_k = \sigma_{k+1}^2 - \sigma_k^2 \quad (36)$$

where it is required that $V_{k+1} - V_k < 0$ for $\sigma_k \neq 0$. However,
it will be shown that $V_{k+1} - V_k < 0$ for $|\sigma_k| > O(T_s^2)$. The
condition $V_{k+1} - V_k < 0$ means that

$$\sigma_{k+1}^2 - \sigma_k^2 < 0. \quad (37)$$

If (27) is substituted into (37)

$$V_{k+1} - V_k = O(T_s^4) - \sigma_k^2. \quad (38)$$

Note that if $|\sigma_k| > O(T_s^2)$ then $V_{k+1} - V_k < 0$. Thus, (38)
shows that σ_k is always converging toward a boundary of
 $O(T_s^2)$ around the desired sliding-manifold and (34) shows that
once σ_k reaches $O(T_s^2)$ boundary it will tend to stay in that
boundary.

282

IV. DISTURBANCE OBSERVER

283 A. Structure of the Observer

284 The structure of the observer is based on (1) under the
285 assumption that all the plant parameter uncertainties, nonlinear-
286 ities, and external disturbances can be represented as a lumped
287 disturbance. As it is obvious, y is the displacement of the plant
288 and is measurable. Likewise, $u(t)$ is the input to the plant and
289 is also measurable. Hence, the nominal structure of the plant is
290 defined as follows

$$\begin{aligned} m_N \ddot{y} + c_N \dot{y} + k_N y &= T_N u(t) - F_d \\ F_d &= T_N h + \Delta T(\nu_{in} + \nu_h) + \Delta m \ddot{y} \\ &\quad + \Delta c \dot{y} + \Delta k y \end{aligned} \quad (39)$$

291 where m_N , c_N , k_N , and T_N are the nominal plant parameters
292 while Δm , Δc , Δk , and ΔT are the uncertainties of the
293 plant parameters. Since y and $u(t)$ are measured, the proposed
294 observer is of the following form

$$m_N \ddot{\hat{y}} + c_N \dot{\hat{y}} + k_N \hat{y} = T_N u - T_N u_c \quad (40)$$

295 where \hat{y} is the estimated position u is the plant control input
296 and u_c is the observer control input. If \hat{y} can be forced to track
297 y , then the control input to the observer becomes $T_N u_c = F_d$,
298 what can be easily verified by determining the value of the
299 equivalent control for system (39), (40) in manifold (41). From
300 the structure of it follows that control input to the observer
301 u_c consists of the terms related to hysteresis effects ($T_N h +$
302 $\Delta T \nu_h$), the terms related to the PZT parameters uncertainties
303 ($\Delta m \ddot{y} + \Delta c \dot{y} + \Delta k y$) and the term related to the uncertainty
304 in the conversion parameter ($\Delta T \nu_{in}$) thus estimating total
305 disturbance [as defined in (39)] but not the components of the
306 disturbance separately. The observer controller that is used is
307 in the sliding-mode-control (SMC) framework. Selecting the
308 following sliding manifold

$$\sigma_{obs} = \lambda_{obs}(y - \hat{y}) + (\dot{y} - \dot{\hat{y}}) \quad (41)$$

309 where λ_{obs} is a positive constant. If σ_{obs} is forced to zero
310 then \hat{y} is forced to track y . It is known from the analysis in
311 the previous section that condition of the same form as (10)
312 $\dot{\sigma}_{obs} + D_{obs} \sigma_{obs} = 0$ guarantees $\sigma_{obs} \rightarrow 0$. If (41) is plugged
313 into $\dot{\sigma}_{obs} + D_{obs} \sigma_{obs} = 0$ then

$$(\ddot{y} - \ddot{\hat{y}}) + (\lambda_{obs} + D_{obs})(\dot{y} - \dot{\hat{y}}) + \lambda_{obs} D_{obs}(y - \hat{y}) = 0 \quad (42)$$

314 where D_{obs} is a positive constant and it can be seen that the
315 transients of the closed-loop system are defined by the roots
316 $-\lambda_{obs}$ and $-D_{obs}$. The controller that will be used in the
317 observer is the same as the controller defined by (18). From
318 structure (40), it can be seen that the input matrix B in (18) is

$$B = \begin{bmatrix} 0 & -\frac{T_N}{m_N} \end{bmatrix}^T \quad (43)$$

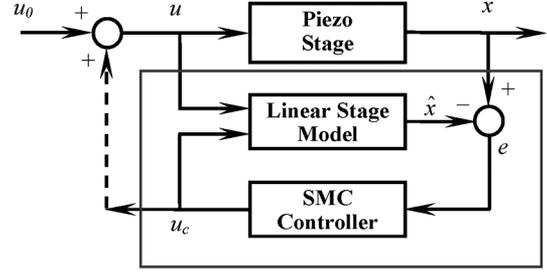


Fig. 2. Observer implementation.

and the matrix G in (18) for this case is

$$G = [\lambda_{obs} \quad 1]. \quad (44)$$

Thus, after some simplifications, the controller can be

$$u_{ck} = u_{ck-1} - \frac{m_N}{T_N} \left(D_{obs} \sigma_{obsk} + \frac{\sigma_{obsk} - \sigma_{obsk-1}}{T_s} \right) \quad (45)$$

here

$$\sigma_{obsk} = \lambda_{obs}(y_k - \hat{y}_k) + (y_k - y_{k-1})/T_s - (\hat{y}_k - \hat{y}_{k-1})/T_s. \quad (46)$$

The observer implementation is best described by Fig. 2. Positive feedback of u_c would, ideally, force the system to behave close to an ideal system defined by

$$m_N \ddot{y} + c_N \dot{y} + k_N y = T_N u_0(t) \quad (46)$$

where $u_0(t)$ is the uncompensated control input to the system. However, this is just the ideal case and in reality the dynamics of the observer would lead to differences between the real disturbance and the estimated disturbance.

B. Observer Dynamics

As it was mentioned previously, the dynamics of the observer has to be analyzed in order to see how close it is possible to force the system to behave ideally as defined by (46). Consider the state-space description of (39) and assuming that the disturbance F_d is matched

$$\dot{x} = Ax + Bu - Bd \quad (47)$$

where $F_d = Bd$, and the matrices A and B are given by

$$A = \begin{bmatrix} 0 & 1 \\ -k_N/m_N & -c_N/m_N \end{bmatrix} \quad \text{and} \quad B = \begin{bmatrix} 0 \\ T_N/m_N \end{bmatrix}. \quad (48)$$

The discrete-time counterpart of (48) is

$$x_{k+1} = \Phi x_k + \Gamma u_k - \Gamma d_k \quad (49)$$

where the matrices Φ and Γ are given by

$$\Phi = e^{AT_s} \quad \text{and} \quad \Gamma = \int_0^{T_s} e^{A\tau} B d\tau. \quad (50)$$

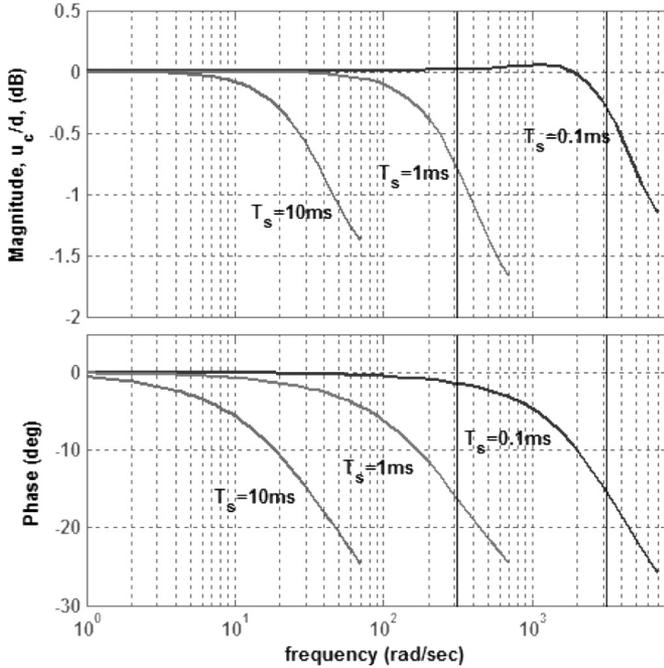


Fig. 3. Frequency response of estimated disturbance w.r.t. disturbance.

338 The disturbance observer is also of the form

$$\hat{x}_{k+1} = \Phi \hat{x}_k + \Gamma u_k - \Gamma u_{c_k}. \quad (51)$$

339 If (51) is subtracted from (49) then

$$e_{k+1} = x_{k+1} - \hat{x}_{k+1} = \Phi e_k - \Gamma (d_k - u_{c_k}). \quad (52)$$

340 The discrete-time transfer function of e_k can be found from

$$e_k = -(I \cdot z - \Phi)^{-1} \Gamma (u_{d_k} - u_{c_k}). \quad (53)$$

341 Similarly, the controller defined by (18) can be written in
342 transfer function form

$$(1 - z^{-1})u_{c_k} = -(GBT_s)^{-1} ((1 + DT_s) - z^{-1}) \sigma_{obs_k}. \quad (54)$$

343 If $D = 1/T_s$ and (54) is simplified further

$$u_{c_k} = -\frac{(GBT_s)^{-1}(2z - 1)}{z - 1} \sigma_{obs_k}. \quad (55)$$

344 Note that $\sigma_{obs_k} = Ge_k$; therefore, using (53) and (55)

$$u_{c_k} = \frac{(GBT_s)^{-1}(2z - 1)G(I \cdot z - \Phi)^{-1}\Gamma}{(z - 1) + (GBT_s)^{-1}(2z - 1)G(I \cdot z - \Phi)^{-1}\Gamma} d_k \quad (56)$$

345 From (56), it is possible to analyze the sensitivity of the
346 disturbance observer w.r.t. disturbance. Fig. 3 shows the fre-
347 quency response of the observer estimated disturbance w.r.t.
348 disturbance for cases when the sampling-time is 10, 1, and
349 0.1 ms. For the observer characteristics shown in Fig. 3, the
350 controller parameters are as follows: $D_{obs} = \lambda_{obs} = 1/T_s$.

351 It will be interesting to see the effect inclusion of disturbance
352 compensation has on the overall closed-loop system.

C. Closed-Loop Performance With the Disturbance Observer 353

In this section, the sensitivity of the controlled position with
354 respect to disturbance will be analyzed. Consider (49), the
355 open-loop transfer function can be written as

$$x_k = (I \cdot z - \Phi)^{-1} \Gamma (u_k - d_k). \quad (57)$$

For simplicity, (57) will be written as

$$x_k = H_{OL}(z)(u_k - d_k). \quad (58)$$

Similar analysis can be done for the controller defined by (18),
358 which can be written as

$$u_k = (GBT_s)^{-1} \frac{(1 + DT_s)z - 1}{z - 1} \sigma_k. \quad (59)$$

If $D = 1/T_s$ and (59) is simplified further

$$u_k = (GBT_s)^{-1} \frac{2z - 1}{z - 1} G (x_k^r - x_k) = H_c(z) (x_k^r - x_k). \quad (60)$$

If (60) is substituted into (58) and the estimated disturbance u_{c_k}
361 is added to u_k

$$x_k = H_{OL}(z) H_c(z) (x_k^r - x_k) + H_{OL}(z) (u_{c_k} - d_k). \quad (61)$$

If (56) is written as

$$u_{c_k} = H_{Obs}(z) d_k \quad (62)$$

and substituted into (61) and after simplifications the following
364 result is obtained

$$x_k = H_{CL}(z) x_k^r + H_{Dis}(z) d_k \quad (63)$$

where the transfer matrices $H_{CL}(z)$ and $H_{Dis}(z)$ are given by

$$H_{CL}(z) = (I + H_{OL}(z) H_c(z))^{-1} H_{OL}(z) H_c(z) \quad (64)$$

and

$$H_{Dis}(z) = (I + H_{OL}(z) H_c(z))^{-1} H_{OL}(z) (H_{Obs}(z) - 1). \quad (65)$$

Note that the displacement is $y_k = C x_k$ where $C = [1 \ 0]$

$$y_k = CH_{CL}(z) x_k^r + CH_{Dis}(z) d_k. \quad (66)$$

Now, it is possible to see the sensitivity of the controlled
369 position w.r.t. the disturbance for the case of disturbance com-
370 pensation. Note that if there was no disturbance compensation,
371 then the transfer matrix $H_{Dis}(z)$ would be

$$H_{Dis}(z) = (I + H_{OL}(z) H_c(z))^{-1} H_{OL}(z). \quad (67)$$

Also, note that in the case of open-loop control the transfer
373 matrix $H_{Dis}(z)$ can be found from (58) after including the
374 estimated disturbance u_c defined by (62) with the control input
375 u . This would result with following form of $H_{Dis}(z)$

$$H_{Dis}(z) = H_{OL}(z) (H_{Obs}(z) - 1) \quad (68)$$

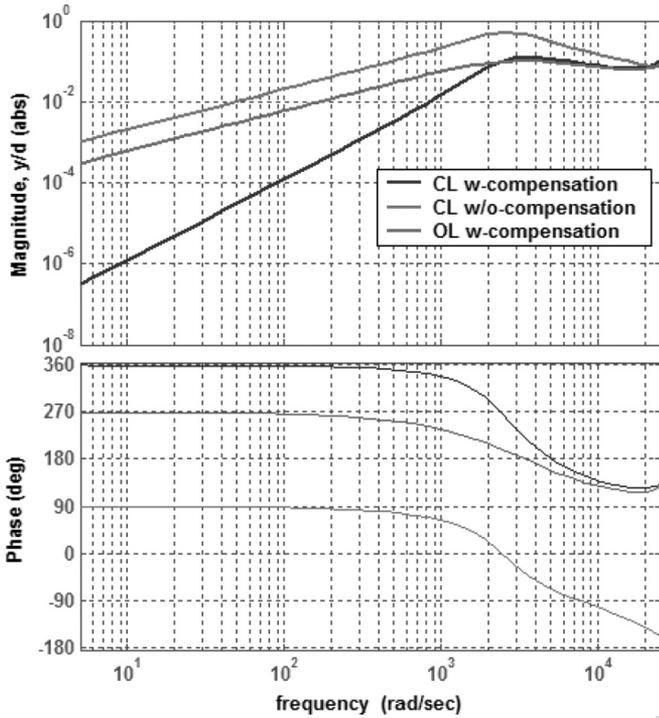


Fig. 4. Sensitivity (micrometer/volt) of the controlled position w.r.t. disturbance.

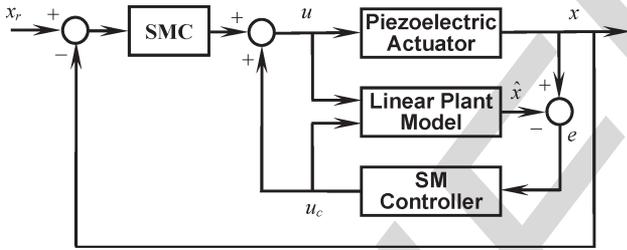


Fig. 5. Control scheme.

377 In Fig. 4 the sensitivity of the closed-loop system for the
 378 cases with and without disturbance compensation are shown
 379 along with open-loop system with disturbance compensation.
 380 Note that when disturbance compensation is included, the sen-
 381 sitivity of the controlled position with respected to disturbance
 382 is less than for the case without disturbance compensation
 383 (Fig. 5). This shows the effectiveness of combining the distur-
 384 bance feed-forward term and the SMC controller. In the used
 385 design, the structure of the disturbance observer is such that
 386 the same controller is used in the main control loop and in the
 387 disturbance observer loop thus simplifying the overall design
 388 procedure.

389 V. EXPERIMENTS

390 In order to illustrate the effectiveness of the proposed control
 391 simulation and experiments are carried out on a single axis of a
 392 three-axis piezostage manufactured by Physik Instrumente sup-
 393 plied by E-664 power amplifier. Table I shows the specifications
 394 of the piezostage. The controller hardware used is the DSPACE
 395 DS1103 with the control algorithm executed on MATLAB and
 396 SIMULINK with real-time link to DS1103.

TABLE I
 PROPERTIES OF THE PIEZOSTAGE

Symbol	Quantity	Value in SI
m_N	nominal mass	1.5×10^{-3} kg
c_N	nominal damping	220 N·s/m
k_N	nominal stiffness	300000 N/m
f_r	resonant frequency	350 Hz
T_N	em transformation ratio	0.3 N/V

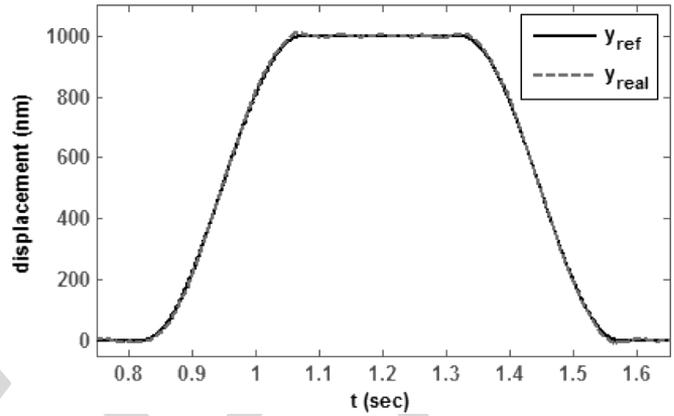


Fig. 6. Open-loop with compensation response to a trapezoidal reference.

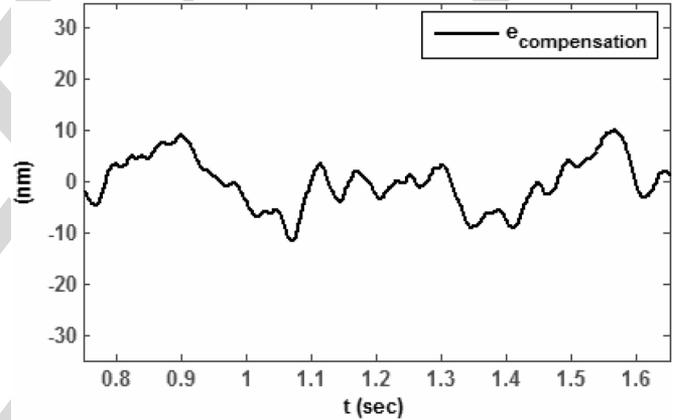


Fig. 7. Compensation error for open-loop with compensation case.

Initial experiments were conducted on the system with just
 open-loop disturbance compensation. Figs. 6 and 7 show the
 response and compensation error with open-loop control. This
 can easily be understood from the results of the sensitivity
 analysis shown in Fig. 4. As it can be seen that, although there
 is no closed-loop controller, the open-loop control with distur-
 bance compensation produces good results as was expected.

Further experiments are conducted with the system with
 closed-loop SMC with disturbance compensation. Fig. 8 and
 9 show the response to a position reference similar to that
 used in the open-loop case. The results show that the proposed
 controller produces good results.

As a means of comparison, the system is experimented with
 PID controller. The results are shown in Fig. 10 and 11. As it
 can be seen, the traditional controller such PID fails to provide
 very good results.

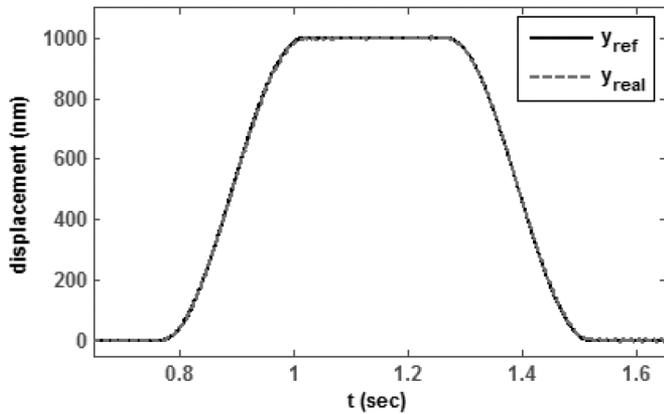


Fig. 8. Response with SMC and compensation.

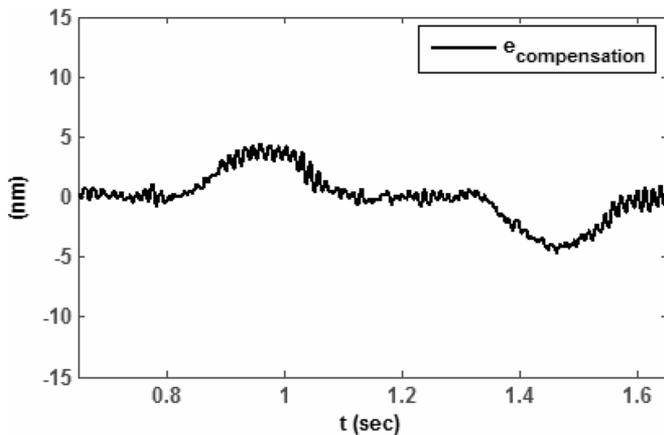


Fig. 9. Tracking performance of SMC with compensation.

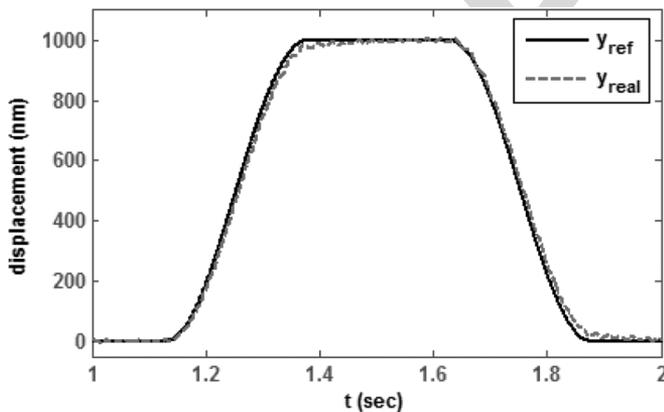


Fig. 10. Closed-loop PID control.

413

VI. CONCLUSION

414 In this paper, the design of a discrete-time sliding mode
 415 controller based on the Lyapunov is presented. The controller
 416 is analyzed for a general system and shown to have very good
 417 performance. It was shown that, similar to [14], the zero-
 418 order hold causes a limitation on the sliding-mode accuracy.
 419 However, it was shown that with partial knowledge of system
 420 dynamics, it is possible to drive the system within $O(T_s^2)$ of the
 421 desired sliding manifold S .

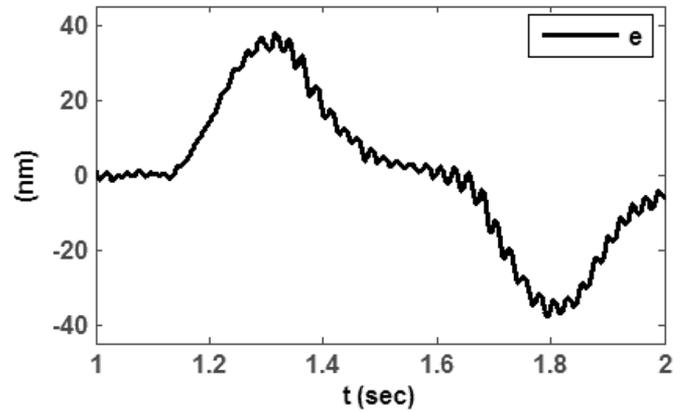


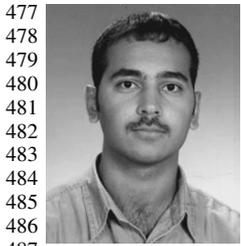
Fig. 11. Tracking performance of PID control.

It was also shown that the introduction of disturbance com- 422
 pensation along with discrete-time SMC in the control of a 423
 piezostage improves the tracking performance. This can be very 424
 useful for applications where simplicity of the controller as well 425
 as high-precision control is required. 426

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